

COMMENTS

Report of the Month

This IO-320 has a problem. Can you tell what it is?

To learn more about where the elements are coming from, click here.

MAKE/MODEL: Lycoming IO-320-E2A FUEL TYPE: Gasoline (Leaded)

OIL TYPE & GRADE: Aircraft Engine Oil OIL USE INTERVAL: 5 Hours

ADDITIONAL INFO: Piper PA18-160, Chrome Cyls

Thanks for discussing this sample with us on the phone. As you know, we're concerned about the amount of chrome this IO-320 is making. To give the engine some credit, chrome did improve nearly 400 ppm compared to the first sample, though the shorter run helped significantly. But at the same time, you only did a partial oil change after the first sample, so a good portion of the chrome could be carryover. We suggest 1. Compression test/borescope of cylinders 2. Looking into why the engine's running hot 3. Another run of 5 hours to see how chrome trends.

	MI/HR on Oil	5		30		
	MI/HR on Unit		UNIT / LOCATION AVERAGES	220		UNIVERSAL
	Sample Date	11/30/2020		8/4/2020		AVERAGES
	Make Up Oil Added			5 qts		
NC	ALUMINUM	17	27	27		8
LIQ	CHROMIUM	219	611	611		5
	IRON	43	129	129		36
2	COPPER	3	9	9		7
Ш	LEAD	1516	4275	4275		2303
Ъ	TIN	1	3	3		1
\RTS	MOLYBDENUM	0	1	1		0
	NICKEL	1	3	3		2
Р	MANGANESE	1	2	2		0
NI S	SILVER	0	0	0		0
	TITANIUM	0	0	0		0
Ě	POTASSIUM	0	1	1		1
Ш	BORON	1	2	2		1
ELEM	SILICON	6	10	10		5
	SODIUM	3	4	4		1
	CALCIUM	5	33	33		17
	MAGNESIUM	1	2	2		1
	PHOSPHORUS	28	104	104		844
	ZINC	2	6	6		4
	BARIUM	0	0	0		0

The owner knew something was obviously up when he got that first sample. In addition to metal in the oil report, the engine was seeing consistently high oil temps in cruise (210-230°F), and high CHT temps in #1 & #3 cylinders, as well as slightly rough running/vibrations in cruise & climb. Additionally, he reported, the engine felt very underpowered. "My engine is supposed to be rated at 160HP, but we had quite a difficult time getting off the water with two passengers, and one instructor who flew it said it felt like it had about as much power as his 100HP super cub. (His plane however is not on floats.)

I suspected the high oil temps might be from a poorly designed/located oil cooler, but several mechanics said that's likely not enough to cause the consistent high temps I was getting, especially since we had the issue when flying in colder temperatures. Since I bought the aircraft used and didn't know the complete history on it, and since the engine only had ~225 hours on it in the 20 years since the major overhaul was done, I decided that I'd play it safe and pull all the cylinders after getting your oil analysis results of the extremely high chrome content, twice in a row.

The company who is inspecting and fixing/replacing my cylinders called me yesterday saying the following was wrong with my cylinders: the wrong piston rings were installed for chrome cylinders; heavy wear in every cylinder- scratches/gouges in #1 & #3 which tells him the plane likely sat for over a year in between startups; evidence of blow by in #1 & #3 - those cylinders also had small cracks developing; and several valves installed backwards (intake installed on exhaust side, & vice versa). He said this isn't a huge issue, but shows the laziness or inattention to detail by the mechanic who installed them (aka a warning sign that they did other things wrong, like use the completely wrong piston rings). So I'm having to get 2 new (reman'd) cylinders to replace the ones with cracks. And then he's overhauling everything else."



Report of the Month

This O-300 has a problem. Can you tell what it is?

To learn more about where the elements are coming from, click here.

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MAKE/MODEL: Continental O-300-A Gasoline (Leaded)

OIL TYPE & GRADE: Phillips XC (A/C) 20W/50 OIL USE INTERVAL: 31 Hours

FUEL TYPE: ADDITIONAL INFO: Cessna 172, Mixed Cyl

The owner says: In reference to the oil analysis surprise of a week and a half ago and the loaded oil filter media, we found the source of the aluminum flakes - worn piston pin plugs. This single plug was from cylinder # 4 and two more were worn on cylinder # 2. Also, we think the (front) thrust bearing is worn beyond limits, allowing the crankshaft to move too much. Looks like a major overhaul is coming.

COMMENTS

Thanks for noting the stuck exhaust valve. Aluminum is really high - high enough to show a problem. If temps got high in the cylinder with the stuck exhaust valve, aluminum could be the result. We are also seeing copper, which in this engine is often from piston pin plugs, connecting rod piston pin bushings, or wrist pin problems. With iron, copper can be from the starter adaptor. Maybe all this will improve now that you've freed the valve, but check for problems anyway. Borescope the cylinders and check compressions. Watch for metal.

MI/HR on Oil	31		38	23	21			
MI/HR on Unit	1,079		1,048	1,011	972			UNIVERSAL
Sample Date	1/17/2021	AVERAGES	5/28/2020	10/2/2019	3/31/2019			AVERAGES
Make Up Oil Added	2 qts		4 qts	1.5 qts	1 qt			
ALUMINUM	140	45	10	7	23			8
CHROMIUM	5	5	3	3	10			4
IRON	31	45	19	21	108			35
COPPER	25	14	10	10	10			9
LEAD	912	1658	1374	1380	2965			2146
TIN	1	2	2	1	2			1
MOLYBDENUM	0	1	0	0	2			1
NICKEL	1	1	1	and the owner of the owner.		And Astron	and the second	2
MANGANESE	1	1	0	And Address of the	and the second			1
SILVER	0	0	0	and the state of the	Sector Sector	Carrier and	Re- granter	0
TITANIUM	0	0	0	the states in	ANNI STATE	S. House	and the second second	0
POTASSIUM	0	0	0	South States	and the second			1
BORON	0	1	2	and the second				1
SILICON	8	15	10	100 mg 4		And Contraction of the		7
SODIUM	3	3	3	天神》后,后		S. W. T. V		1
CALCIUM	53	74	62	1. 2 2 2 2 1	and the Low	-T- A	- All States	18
MAGNESIUM	28	20	26	hard a	1 to the second	and the second		10
PHOSPHORUS	87	241	99	the property and	CA SAL SA		1 8 4 A	443
ZINC	27	20	30	2.7	and the second	a state of the	and the second	5
BARIUM	0	0	0	0	0			0

		values					
		Should Be*				 	
SUS Viscosity @ 210°F	93.6	86-105	85.8	88.8	84.3		
cSt Viscosity @ 100°C	18.84	17.0-21.8	16.97	17.70	16.59		
Flashpoint in °F	410	>430	480	475	445		
Fuel %	1.0	<1.0	<0.5	<0.5	<0.5		
Antifreeze %	-		-	-	-		
Water %	0.0	0.0	0.0	0.0	0.0		
Insolubles %	0.3	<0.6	0.2	0.3	0.5		
TBN							
TAN							
ISO Code							_

* THIS COLUMN APPLIES ONLY TO THE CURRENT SAMPLE

416 E. PETTIT AVE. FORT WAYNE, IN 46806 (260) 744-2380 www.blackstone-labs.com